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GARDEN CITY, LONG ISLAND, N. Y.



AVIATION AND AIRCRAFT JOURNAL

Vol. IX

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American Pilots

THE second Pulitzer Trophy Race was further evidence that American pilots are as good as any in the world.

Foreign authorities have had great shows in the airports of their pilots and some people in this country have received the impression that the United States is outclassed in that respect. It is an absolute fallacy. One of the principal reasons for the lack of mention of individual pilots has been the large number of first class pilots at work in this country. The somewhat unfortunate dearth of aviation news has allowed the daily papers to forget, to some extent, the existence of these men.

The aerial race is an example of the public's forgetfulness of the existence of the pilots employed. It is one thing to fly from one airport to another on a clear warm day, but it is a different proposition to take off in a fog and fly through a succession or two with the temperature below zero. And that is just what the real pilots are required to do very frequently. Flying on schedule in storms requires a much higher order of piloting than residential flying in the neighborhood of an airport. Many of the pilots in the mail service have had over two thousand hours of flying. That figure may not seem much to the layman, but it requires steady flying for a few years to obtain it, or to put it another way, it means about 100,000 miles of travel.

The Army and the Navy have many pilots who can be listed among the best in the world and who are unknown to the general public. The race brought several of these into prominence and there are many more yet to be heard from. The Moscow-Alaska flight was no big feat as the Cincinatti-Cape flight. It was much more of a routine. There were no novelties and the same machines returned without major repairs. In the other famous flights that have been made since the war there is no instance of more than one machine completing its mission. Such public events, as the Pulitzer race allow the average pilot a chance for recognition. It is a matter of military and naval affairs that good work is the duty of an officer and that publicity was not permitted. This is essential to discipline and coordination of effort. But in open competition with airplanes as a sporting event or performing some extremely hazardous enterprise such as crossing the Atlantic or flying to Alaska, it is due to the officer and his men that the fact be known. The public should not believe that because it doesn't know the names of the fine pilots we have, that they don't exist.

The Design of Wind Tunnel Balances

WHEN the first modern wind-tunnel was erected in this country at the Massachusetts Institute of Technology, the National Physical Laboratory type of balance was adopted. In several other tunnels the same balance was adopted without much modification.

The M. P. L. type of balance is an excellent one and has

given the most valuable results. It has been very widely copied in this country because it is difficult to modify a piece of laboratory apparatus with which a man has had no personal experience. But now that a number of men have lived and worked with these balances for a number of years, it would seem as if in future construction a little more originality could easily be shown.

It might be possible to design the recording balances whereby the number of men employed on a test will be diminished. It might be possible to regulate the speed of the wind automatically, and to correlate with the readings of the balances. Stability considerations would be greatly facilitated by an instrument allowing other than the single longitudinal forces to be readily measured.

There is no reason why wind-tunnel balances should become unduly conservative or stagnant.

The Pulitzer Trophy

THE Pulitzer Trophy race has given a new impetus to designers and manufacturing engineers. They have seen the Army and Navy Air Services place a premium on speed and it has occurred them that any amount of money will be spent for the few extra miles per hour.

Races in other sports have been the means of bringing speed to the fore, in horses, automobiles, bicycles and yachts. That 1920 will go into aeronautical history as a year of great advance in airplane design is now assured. The development of the airplane has been noteworthy. The triplane has had its moment. The monoplanes as being used in all sizes and types. From these lines of endeavor will undoubtedly come results which will lead to being in such close a more definite relationship to performance.

Pitot Controls

THE plans that are now being considered for the offering of prizes for contests in the design and performance of aircraft will, if completed, make next year noteworthy in this respect. There have been from time to time suggestions of such contests but only recently have they been based on more than hope.

In the design of airplanes there is the normal competition that is always going on for improvement, but by having a contest for the design alone of a highly specialized type without the hindrance of constructional problems, it is probable that many new features will be brought out.

Controls for commercial types of planes are also being planned, with adequate prizes to make possible real attempts to produce more economical aircraft. That this kind of competition is the most urgent of all is evident.

Commercial aeronautics is now awaiting data which such a contest would produce. A great stimulus would be given to the whole art by a money placed contest which would give real facts as to costs per ton mile of useful commercial load.

The Pulitzer Trophy Race

The Verville-Packard was the Pulitzer Trophy at a speed of 178 m.p.h. for the 100 mile event. This is the same machine that was entered in the Gordon Bennett race and its recent performance shows clearly that it is one of the fastest airplanes in the world, if not the fastest. At a speed test made after the race the machine made 196 miles per hour over a measured mile. It is expected that more speed trials will be made with this machine in the near future and that the world's speed record which is now standing at 312 miles per hour for a straight, open measured mile will be passed by a very margin. The machine was very skillfully handled by Capt. Gordon G. Hooten, who was Maj. Nicholson's alternate in France at the last race. The Packard engine is rated at 680 horsepower.

The second prize was won by Capt. Harold E. Hartney in a Thomas-Morse MB-3 pursuit plane, powered with a 350 Wright engine. Capt. Hartney attained an average speed of 159.5 m.p.h. for the course. His piloting was extremely good and as the first plane to get away he created a great start. Bert Asselt in an Ansaldo Italia gained third place at an average speed of 154 m.p.h.

The race was witnessed by at least 20,000 people. Not all of this number were at Mitchell Field. The results following the course of the race were lined with parked automobiles and large crowds watched the race at both Lafferty and Henry D. Benson Field, the turning points of the course. The absence of serious accidents and the high speeds displayed made a great impression on the spectators. Many who looked at aviation as a race venture, realized its stability and progress when they saw over forty airplanes competing in one event.

The management of the race was excellent and the cooperation of Major Charles, who is in command at Mitchell Field went far toward making the race a huge success. The roll of the officials followed and they are all deserving great praise for their excellent voluntary efforts.

Captain, Commandant
*Benjamin F. Castle,
Chairman

*Caleb S. Bragg,
*Blaney Greenwald,
*David McCulloch,
*H. E. Brown, (Referee)
*F. H. Hays, (Referee)
*Edward O. McGovern,
*Douglas Campbell,
*Alfred E. Hawley,
*Charles E. Merrill,
*Stanley G. Greer,
*Arthur B. Lambert,
*Howard H. Myers,
*W. H. McDonald, Com.
*William A. Lorrain,
*Howard A. Roberts.

Referee
*J. G. Vincent

*Frank Army of Navy
Air Service
*D. S. Hooten, Pilot
*D. S. Hooten, Pilot



THE PULITZER TROPHY

The winning machine in the Gordon Bennett race in France last summer and has various small changes made to it which have vastly improved its performance in regard to reliability. The main dimensions are: Span 28 ft., 2 in.; length 34 ft., 2 in.; height overall, 8 ft., 5 in.; weight empty 2485 lb. and weight loaded 3235 lb. The wing loading is 14.13 lb./sq. ft., and the power loading is 5.65 lb./hp. The plane is fitted with the Packard engine which is rated at 650 hp. at 2600 r.p.m. on premium piston. During the race the engine was not operating at its maximum power but was turning over at 1700 r.p.m.

Jurors

William A. Lorrain

Col. James A. Hays, Jr., Charles B. Warren, Roger D. Williams, Philip A. Cayrol, Philip J. Roosevelt, Charles DeLagrange, W. Belmont Cross, Col. Jefferson Richard Thompson, William C. Patton, Charles Jackson Edwards, Robert D. Waver, Edward H. Thomas, Charles E. Merrill, Alvin H. Hawley

Referee

Caleb S. Bragg

Engine

Richard F. Hays, R. A. Lorrain

Technical Committee

Henry Cross, Chairman

James Brown, James G. Hooten, G. C. Loring, Major V. E. Clark, Alexander Klenka, Charles M. Maury, L. B. Gossman, Victor, Edwards, George Maury, Mayo D. Brown, Charles Lawrence, Elmer A. Sperry, C. Wilbur, Charles Wright, Col. Jesse H. Vincent, W. L. Brown.

Army Ordnance Commission

Captain Arthur Clardy, Commandant of Mitchell Field, Capt. Henry Allen, Jr., Charge of Personnel, Louis Gossman, Adjutant, Mitchell Field, Colonel Brown, Charge of Medical Research Laboratory, Captain Langston, in Charge of Flying Field, Major Chas. Post, Surgeon, Captain Smith, Assistant, Charge of Flying Field, Major Strong, Flight Surgeon, Lt. Rankin, Patrol



ONE VIEW OF WINNING VERVILLE-PACKARD
Photograph by U. S. Army Air Service

and an infinitesimal mistake in control might easily result in disaster. It has been often said that a pilot may be expelled but it requires an excellent pilot to prove it. The winning of the race demonstrates the easiness of both the pilot and the plane.

The winner of second place was Capt. H. E. Hartney in a standard military pursuit machine with only half the engine power of the winner. The Thomas-Morse MB-3 with a Wright 300 hp. engine has been in development for two years and the machine entered in the race was not even on any way. One of the rules of the contest was that the parts of planes should be of the same kind to be standard parts and could be replaced by a similar part taken from a similar machine without affecting the control or operation of the machine.

The general specifications of the Thomas-Morse MB-3 are as follows: Span 28 ft., 2 in.; length 34 ft., 2 in.; height overall 8 ft., 5 in.; weight empty 2485 lb.; weight loaded 3235 lb. The wing loading is 14.13 lb./sq. ft., and the power loading is 5.65 lb./hp.

The machine is of particularly pleasing appearance and gives an impression of speed when in flight. This feature was enhanced by the skillful piloting of Capt. Hartney who flew an extremely straight course and made very good turns at the pylons. It is toward to turn only the most accurate of the stick immediately. It shows that the American Air

Service has reached an extremely high level when a standard American built pursuit plane powered with a stock American engine can make such a fine showing against such a large field.

The winner of third place was Bert Asselt in the Italian Italia fitted with a SPA 120 hp. engine. While this plane is of Italian manufacture it is owned and was operated by Americans. The official title of this machine is Ansaldo I and it was made by the manufacturer of the SPA machine. Asselt's flying was up to his usual standard and he gave a pleasing performance while sweeping up and losing his machine before crossing the starting line.

Next, St. Clair Street, the commander of the Italian Albatross flying expedition was in fourth place with an Albatross model D pursuit plane fitted with a Wright 300 hp. engine. This is another example of a standard Army service plane showing up very well in a competition. The Albatross is one of the standard planes adopted by the Army Air Service as a service plane and was flown without alteration.

The fifth place was won by Leach (Jr.) A. Lacroix in a Vought VE-7 with a Wright 180 hp. engine entered by the Navy. It is the outstanding other machine in the Vought class and so was the prize for that type. His home was perfect and he was always within one hundred yards of the pylons when rounding the corner.



COURSE OF THE PULITZER TROPHY RACE

This victory is very gratifying to American engineers as it is a vindication of their ability to make out the world's best designs. The experience gained in the design and operation of this machine will doubtless tend to improve the standard pursuit machines.

The piloting of Capt. Hartney certainly has to take rank with the world's most difficult pilots. The difficulty of handling as a surprise machine nearly at the speed of the maximum speed. The Verville lands at a high speed



A THOMAS-MORSE MB-3



SIDE VIEW OF THE TAIL FIN OF JORDON

The winners of the class prize were as follows:

DE HAWKLAND CLASS

- 1.—Lieut. John P. Keston, U. S. A.
- 2.—2nd Lieut. C. E. Keston, U. S. A.
- 3.—2nd Lieut. J. B. Wright, U. S. A.

VENTURE CLASS

- 1.—Lieut. [2d] A. L. Laverne, U. S. N.
- 2.—Lieut. [2d] W. B. Geyne, U. S. N.
- 3.—2nd Lieut. L. H. Sanderson, U. S. M. C.

SE-5 CLASS

Capt. Maxwell Miller, U. S. A.

CHALLENGER INVENTION PRIZE

- 1.—Capt. H. E. Hartney, U. S. A.
- 2.—Lieut. St. Clair Smith, U. S. A.

In cases where there was a monetary prize which was won by a service pilot, the money goes to the relief fund of the Army or Navy and the pilot received a trophy. This is because Army and Navy officers are forbidden by law to be paid for their services in any way.

The Navy entered a Loening monoplane with wings folded specially for speed and after covering almost the entire course at a speed of about 155 m.p.h. The pilot, Lieut. H. G. Tinsley, U. S. M. C., was forced to land about a mile from the finish to avoid freezing the engine. A water connection broke while the machine was on the third lap but Lieut. Tinsley kept on and was able to keep up the pace nearly long enough to finish. It is greatly to be regretted that he landed before the engine came instead of trying to cover the last mile.

One somewhat serious incident of the race was the entry of the last mile of the flying machine. Its own white paint and red cross insignia caused a great deal of comment from the spectators.

In the large crowd viewing the race there were numerous prominent officers and civilians. General Peckham and General Bessie were both present, and the Army Air Service was further represented by Maj. Gen.

C. T. Mosley, Chief of Air Service, Brig.-Gen. William Mitchell, assistant to the Chief of Air Service, who was the first to reach and congratulate Capt. Mosley. Capt. T. T. Green, the chief of Naval Aviation, and Air Commander L. E. O. Charlton, British Air Attache, were interested witnesses of the race. Another foreign representative was Anthony H. G. Fisher, the famous Dutch airplane designer.

In charge of the Navy machines and pilots was Lieut. Com. V. C. Griffin, Jr., of the Atlantic Fleet Shipping Division, who was detailed to this duty by Admiral Henry B. Wilson, commander-in-chief of the fleet, at the request of Capt. T. T. Green, director of Naval Aviation.

A committee in charge of Lieut. Com. M. C. Chase, suggested the Navy planes two days prior and on the day of the race it was that they were up to specifications and in the best of condition.

The Bureau of Engineering was represented by Lieut. Com. S. M. Kress and George W. Miller, Bureau of Construction & Repair by Capt. J. C. Blumher and Lieut. Com. O. F. Felt; the Bureau of Navigation by Lieut. Com. R. M. Griffin and Lieut. W. L. Richardson, photographer. Com. G. C. Westcott and Com. H. C. Richardson, both of the Construction Corps came from the Naval Aircraft Factory.

Col. Thurman H. Bane represented McCook Field while Col. Wm. E. Gifford and major officers on duty in Washington witnessed the race. Among the spectators present were noted, Alfred E. Verville, William A. Brock, Elmer H. Curtis, Orville C. Leasing, Walter Phelps, Charles Vaughn, E. D. Thomas, George L. Macdonald, Jr. and

particularly all the personnel concerned in the present aeronautical talent of the country was standing the contest.

Col. Jesse G. Vossard, Charles Lawrence and George Mead were both present after the engine failure of the race. Mr. prominent in the industry in all its branches was extended spectators while New York apartment was to be seen in all parts of the field.

As the contest was



AIR SERVICE FORDNEY MACHINE

regarded as a speed contest the length of the course has not been accurately determined and a survey is to be made to get the exact distance. It is thought that the course is more than 120 miles in length, the speed of the machines will be proportionately increased. As it is the speed of the winner is probably a record for a closed circuit. The winner of the Gordon Bennett race in France made an average speed of 100 m.p.h., nearly ten miles an hour less than that made by Capt. Mosley.

Several other machines created a great deal of interest by their performance. They were the two Curtiss biplanes, the Remondy biplane and the Mooney-Bushnell type Air monoplane. The whistling noise made by the triplane caused many of the crowd to refer to them as "Whistling Bikes." These planes experienced engine trouble or they would have probably been up among the winners. The Mooney-Bushnell gave an

exhibition of its flying qualities after the race that attracted a considerable crowd and added a great deal to keeping up the interest after the finish of the race. All the machines on the field were an exhibition. The L. W. F. Fordney was added out of her harness and afforded the spectators an example of what the United States can do in building large machines. The Messenger which was designed also by Verville was flown by General Mitchell. Its top was emphasized when near the goal.

The special train of the Long Island Railroad was pooled and helped in no small degree in making the event a success. It is New York's contention that it has no airport that is so easily accessible. Had there been one for the race, three would have been no need of special trains and many more people might have come from the city, especially if the airport was located near some rapid transit system.

The Finishing Order and Time

FORM- TYPED NO.	PILOT	AIRPLANE	ENGINE	PLANNED TIME
1	Capt. C. C. Mosley, U. S. A.	Verville	Packard 600	44:29.57
2	Capt. H. E. Hartney, U. S. A.	Thomas-Morse	Wright 300	47:30.63
3	Albert Arita	Ameside 1	B. P. A. 350	51:05.05
4	Lieut. St. Clair Smith, U. S. A.	Wright 300	Wright 300	52:01.61
5	Lieut. A. L. Laverne, U. S. N.	Vaughn	Wright 160	53:39.15
6	Lieut. J. P. Keston, U. S. A.	De Havilland	Liberty 400	56:05.58
7	2d Lieut. C. E. Keston, U. S. A.	De Havilland	Liberty 400	56:09.26
8	Lieut. J. B. Wright, U. S. A.	De Havilland	Liberty 400	56:52.26
9	Lieut. C. Cunningham, U. S. A.	De Havilland	Liberty 400	57:04.12
10	Lieut. D. L. Cudley, U. S. N.	De Havilland	Liberty 400	57:40.76
11	Capt. Harvey S. Hines, U. S. A.	De Havilland	Liberty 400	58:05.26
12	Lieut. D. C. Finch, U. S. N.	De Havilland	Liberty 400	58:32.94
13	W. H. P. Taylor, U. S. A.	S. P. A. 350	S. P. A. 350	59:35.59
14	Capt. Maxwell Miller, U. S. A.	SE-5	Wright 180	59:42.67
15	Lieut. W. B. Geyne, U. S. N.	Vaughn	Wright 160	59:51.03
16	Lieut. L. Claude, U. S. A.	De Havilland	Liberty 400	59:56.71
17	Lieut. L. H. Sanderson, U. S. M. C.	Vaughn	Wright 150	59:59.26
18	Lieut. W. H. Lawrence, U. S. A.	De Havilland	Liberty 400	59:59.96
19	Capt. H. S. Miles, U. S. M. C.	De Havilland	Liberty 400	59:59.94
20	Lieut. Louis Davis, U. S. A.	De Havilland	Liberty 400	61:15.87
21	Lieut. W. B. Brown, U. S. A.	Vaughn	Wright 160	62:23.28
22	Lieut. T. Moffat, U. S. A.	Vaughn	Liberty 400	61:47.45
23	Lieut. J. E. Isaac, U. S. M. C.	Vaughn	Wright 300	61:54.36
24	Capt. M. Canby, U. S. A.	De Havilland	Liberty 400	62:25.25
25	Charles Colt	Mooney-Bushnell	De Havilland	72:34.33

Notes on the Pulitzer Air Race

by Alexander Klemm
Consulting Aeronautical Engineer

One can hardly imagine a better method of arousing general public interest, stimulation, of inducing the same general public, of stimulating scientists and designers to greater efforts than an act of the character of An inspection of so many varied and brilliant types at one time is almost a valuable expert lesson, to the designer. The process of tracing up the plans and the region to their very highest price, and the effort to secure over an hour's reliability at maximum speed is also a very valuable practical experience, particularly as regards reliability of the power plant.

From a strictly technical and analytical viewpoint, a race of this character is far less valuable however than the careful performance testing of isolated phases. In a regulation performance test each factor is the skill of the pilot in making accurate turns and in keeping to the correct route, and the effect of wind and weather are largely eliminated. The most careful record is also made of the exact condition of the aircraft at the particular of the given test carried out. In the hurry and confusion of a race for a new record factors are naturally lost sight of, and even technical considerations much more difficult of not responsible.

Reliability in the News

Of the sixteen forty-fives were ready to start, thirty seven started, thirty-three on schedule, nine twenty-five finished. Considered the desirability of using woodchips, this is an excellent showing. Of the thirty, twenty-five were not complete the first. One was damaged for an infestation, not the regulation, while seven experienced mechanical troubles in the power plant, such as broken water connections, etc. This emphasizes once again the structural reliability of the modern airplane, and also the necessity of greater reliability in the power plant.

Improvement in Speed from Standard to Rumba Machine

For those looking forward to participation in further work of this character, perhaps one of the most interesting things to know is how much better speed can be got out of a machine when stripped for a race, than when carrying its regulation load and accessories.

Although this will vary largely for each ship depending on the total load which is carried on the standard machine, on the amount of assistance needed equipment and on the ability of the motor to be driven at speed for an hour at a rate, instead of the few minutes on a performance test, the following comparative table of the first six machines may be useful for approximate reference. In this table loss of time in making turns and so forth must of necessity be neglected, and the distance of 150 miles taken as a basis.

It can be seen at once from the accompanying table that

PLACE	ENGINE	RATED HORSE POWER	NORMAL WEIGHT WEIGHT	NORMAL WEIGHT PER SQ. FT.	NORMAL WEIGHT PER HORSEPOWER	WEIGHT LEADY	MAINTENANCE EFFICIENCY PERCENT	MAINTENANCE EFFICIENCY WHEELING IN LBS.
Yerville-Packard	Packard	620 hp. at 2000 r.p.m.	3203	34.12	5.96	2485	17	17
Thomas-Morris Single Screw Engine	Thomas	620 hp. at 1800 r.p.m.	3099	8.3	6.2	1505	152	18
Amstar 10 Horse	S. P. A.	220 hp.	3866	8.3	8.0	3470	337	18
Greene Single Screw Engine	Greene	320 hp. at 1800 r.p.m.	2642	8.3	7.5	1770	187	18
West 10 H.P.	West	380 hp. at 1800 r.p.m.	3461	11.25	8.25	2164	216	18
Four De Havilland	Liberty	400 hp. at 1600 r.p.m.	2908	8.2	8.0	2230	136.5	18

while for the high speed machines such as the Thomas-Morse and Ansaldo roughly seventeen miles an hour have been added above the normal speeds. In the case of the Vought nearly thirty miles have been added, and in the case of the De Havilland at least fifteen miles per hour.

In some of these machines as far as is known, have the wings been cut down in area. These results have evidently

been obtained by decreasing the loads carried and removing all resistance forming objects such as guns, sights, gun mounts, armoured indicators and so forth.

It would appear that in the case of very fast machines where the last possible refinement has already been introduced much additional refinement is out of the question, but in the case of comparatively slow machines there is much greater scope for speed increase.

Prior to the war, it was the secret of each contestant who special methods he adopted. It would be interesting to know now that the spirit of contest is over, what exactly was done in all cases. Some are fairly obvious.

Outing down the wing area is the simplest yet most effective device. These seem to be exactly my fault in the direction, with the ever growing skill of the pilot in looking at high speed, and the fact that in a race climb is of im-

consequence. A high pitch propeller may readily be substituted for the average all round stick, in which case the main aim is to be given to climb and riding on with a speed. The setting of the wings may generally be decreased, particularly if they are set at an angle of 3 to 5 degrees on the standard plane. With a fast machine, in which stability has been secured by using a large dihedral between the wings and the tail surface, this dihedral may be diminished with perceptible results.

All useful land except the plot and fact unfavourable for them can be disposed with. All gullies showing resistance can be removed. A suitable covering over control surface gully may be helpful. The maximum possible compromise of the region, a carburetor sitting suitable for ground level a full throttle, with an increase in radiator area, are simple solutions appropriate for the power plant. For each machine some special operation is always to be found.

The Ferville-Parkland

The Verrill-Packard is the winning plane, actually earned great interest. It has a span of 26 ft., 2 in.; crown height, 24 ft., 2 in.; weight empty, 2500 lb.; weight fully loaded, 3500 lb.; wing loading, 130 lb. per sq. ft.; wing area, 26.7 sq. ft.; maximum speed, 141.2 ft. per sec. It is the most rapid at 500 ft. and is leading per horsepower between 5.00 ft. . . . Allowing for the fact that the machine was more lightly loaded in the race than in actual service, it is surprising to see how readily the packard can be loaded to its maximum weight. The machine is not a high lift plane. If such excessive wing loading can be readily handled, designs of single-engine aircraft will more readily be coming down of wing area to secure greater speed. The Verrill-Packard did not secure the world's record, but it is a very fine machine and is a very good all-rounder for the very low loader per horsepower—a class

THE TWO MEN RESPONSIBLE FOR THE DESIGN OF THE VEPRILE-PALMERS PLANE AND ENGINE. COL. JAMES G. VEPRILE AND
ALBERT T. PALMERS.

Edward Weiss is well-known in the neurological world. He is the associate in the Air Service in connection with the present plant program studies, another work. He is a member of the Society of Neurological

about V. Varadin, the designer of the winning abstract of the Fathoms the waylens industry in 1964 with the Engineering Department of the University of the Western Australia. He worked for General Motors Corporation and the Western Australia. He worked for General Motors Corporation and the Western Australia. He worked for General Motors Corporation and the Western Australia.

in 1949 he was sent abroad by the Director of the Bureau of Aeronautics to study the aircraft design of the V-2, which was then in the hands of the Germans of the Luftwaffe. He was sent to the V-2 design office in Peenemünde, Germany, where he was assigned to the design of the V-2, which was then in the hands of the Germans of the Luftwaffe. He was sent to the V-2 design office in Peenemünde, Germany, where he was assigned to the design of the V-2, which was then in the hands of the Germans of the Luftwaffe.

gradually into the venter fasciage. The use of venters allows the fin to be built enclosed with the fasciage, and everywhere as at junction of wings with fasciage, and stanchion wall fasciage facing is finer employed, while the lower of the fasciage fair will behind the delivering aerodynamical pressure of the corbels.

The single biphasic I-drug has again been used successfully for a large, fast and heavily loaded machine. The arrangement of the lift wires, as shown in the photograph, by means of which the lift wires are made to serve as drag and drag wires, probably helps to strengthen the brass connection. The S type form and all strut housing at the back of machine and system.

The Thomas-Moran Single Sister Scout

The M&B report shows that great care has been taken to design to eliminate structural weaknesses. Fittings are placed inside the wings and fuselage wherever possible. Aluminium wires are used as all exposed fasteners. The wing bracing is somewhat out of the ordinary. Left wings are brought from the outer struts to the fuselage instead of to the inner struts, thus allowing the use of smaller struts in the inner struts. The shape of the fuselage is borrowed at the nose by lower wing sections. The design is light on wall thickness and enables a service ceiling of over 33,000 ft. to be reached.

Grip in elbow and axpect zones are both small, apparent without serious loss of aerodynamic efficiency. The span is 28 in and length around 19 in giving with some small dimensions the much desired ease of handling. A chord of 5 ft. 3 in. on both wings and a gap of 4 ft. 7 in. with no struts are used.

The emphasis which has been placed on the performance of this entry as a stock pursuit plane now being manufactured in quantities for the Air Service is justified by its success.

The Double-Edged

This single-engine, one of the few entered by a private firm, is a single streamer of very clean design. The wing span is only 12 ft., one of the smallest in the race. A six-cylinder vertical S.P.A. motor completely covered and



THE TWO MEN RESPONSIBLE FOR THE DESIGN OF THE VEPRILE-PALMERS PLANE AND ENGINE. COL. JAMES G. VINCENY AND
ALBERT T. THURMAN.

ended by a cone radiator across the front end of 12 feet long and gives the latter a deep narrow section with 6 sides. Aft of the cockpit the lines flare into a triangular section with the apex at the bottom.

This is the characteristic Arnoldo shape and gives the impression of a very sharp plain flame which varied from below. The fuselage is covered with veneer and waxes are eliminated by the use of plywood bulkheads.

Wing housing is of the conventional single-strut type with double cables and fairing between them. The lower plane struts are of steel, tapered, and the upper control are of the rigid type so that there is little structural distortion of the wing.

The Drone D
The Drone, type D was a highly creditable entry, as finished the race with an average speed of 182 miles per hour. It was in no sense a stripped or lightened ship, and needs no fuel equipment except the gas.

[illegible]

The Granco Type D is conventional in construction, but very neat and well finished. The elevator is in one piece as the rubber is clear above the horizontal turn surface. In a standard form it has a maximum speed of 147 m.p.h. and climb of 10,000 ft. in 5.0 minutes.



A DUE-ARM, SEVERAL OF WHICH WERE OF THE RACE

The Vaught and the DeHavilland

The Vaught VE-7 and the DeHavilland are so well known as to call for no special comment. The VE-7 is a particularly clean training machine with a Wright model E engine rated at 180 hp. The surprising speed of 143 m.p.h. attached to the plane shows very good design for a two-seater. Streamline wires are used to good effect. The wing loading of 36 lb. per sq. ft. with normal full load is 7.35 lb. per sq. ft., probably the lowest in the race, while the normal gross loading of 13.6 lb. per sq. ft. is probably the highest. Engine wings each 38 ft. span and 4.62 ft. chord with 4.67 ft. gap have the conventional radial wing tips.

The DeHavilland is more heavily loaded and powered than the Vaught and has a normal speed 11 m.p.h. greater than the VE-7, not adverse designs of the design which could not be charged for the race limited the improvement in its performance so that it was delayed by an appreciable margin to the rear. Greater loadiness was undoubtedly a factor in favor of the Vaught in making the strenuous turn required at the race.

Comparison with the Gordon Bennett Winner

In conclusion it is interesting to compare the results of the Verdille-Parkard with the winning machine of the Gordon Bennett. Leonard's Newport weighs fully loaded 2690 lb. or 10.6 lb. per sq. ft. With a 140 hp. engine the power loading is 0.58 hp. per sq. ft. The Verdille figures are 14-12 lb. per sq. ft. and 0.66 hp. per sq. ft. based on 830 hp. at 2000 r.p.m. It is unfortunate that the carburetor was such during the race that the Parkard 540 two or three headed revolutions short of 2000 and it is therefore difficult to compare the power loading of 0.66 based on 400 hp. for the Parkard with 0.58 lb. hp. for the Newport. Nevertheless, the American machine was doing 175 miles per hour in this race as compared with the 168 miles per hour of the Newport in the Gordon Bennett.

Trials for the Speed Record

Captain Corlies C. Messler, in the United States army's Verdille-Parkard biplane, brought the American speed record up to 184 miles an hour in a series of test flights at Mitchell Field, Mineola, L. I., Saturday, Nov. 27, but failed to establish a new world's record. In order to bring the record round to this country, the present plane will have to add another six miles an hour to this speed.

Although the plane failed to realize the expectations of some officials, the flight brought to light a few interesting problems which, when solved, may bring to this country premier speed honors of the world.

During each of the six test runs it was observed that upon the ground the powerful 560-horsepower engine was turning

over at a rate of 3,356 revolutions a minute. The moment the airplane took the air, however, the rotor speed rapidly fell, until it was 200 revolutions a minute lower than the ground speed. Careful adjustment of the carburetor to overcome this condition at the end of each test failed to produce any different results.

Speed Greater Factors

Col. J. O. Valiant, designer of the engine, who was watching the experiments, after careful observation expressed the opinion that the terrific speed resulted in a vacuum being created at the back of the carburetor, which prevented that instrument from performing the proper gas mixture to insure the full power of the engine.

While the general design of the remarkable airplane is very clean, the tests also showed that further streamlining can be accomplished, particularly about the engine cooling and the windshield behind the pilot's cockpit.

Incidentally after the conclusion of the tests, it was ascertained that the plane would be shipped to McCook Field, Dayton, Ohio. There the airbomber will be placed in the experimental wind tunnel and subjected to the approximate air conditions for a high speed flight. Careful observations will be made during the test in an effort to solve the mystery surrounding the performance in the recent flight.

When this has been done the plane will be sent over a measured course again for the purpose of winning for America the speed record of the world. Technical efforts of the army are working fast that if the carburetor trouble is overcome this machine will make 300 miles an hour.

Thomas-Morse Shows Speed

Although the army air service failed in its major expectation, the service achieved signal satisfaction in the wonderful performance of a stock Thomas-Morse speed equipped with a 300-horsepower Wright engine. This is the identical machine that was second place in the Thanksgiving Day race. Capt. Harold E. Hartney, the famous American ace, was the pilot on both occasions. In tests he succeeded in getting the machine which was designed for a speed of 150 miles an hour up to 171.25 miles an hour over the measured course.

Captain Hartney covered the one-kilometer course in 39.3 seconds. Thomas-Morse 300-horsepower plane in 31.02 seconds. The average for any time over the 1/2-mile course on Thanksgiving Day was 166.6 miles an hour, while the average for six test flights was 178.4 miles an hour. Thus, it is expected, will prove a new world record for the type of plane. At the conclusion of the flight the same machine was taken up by Capt. St. Clair Harty, of Alaska flight group. While at a low altitude the machine went into a nose dive. It was at this altitude that Capt. Harty brought the machine out and as it was about to crash.

Fokker Commercial Airplanes

The success of the Fokker sailplane flying monoplane led to the design of a machine of similar type for commercial purposes. In order to keep the running costs of this essentially unserviced machine as low as is at present possible, the engine power was kept down to a maximum of 250 hp. Resulted

To avoid the usual spilling about of machine while filling up the tanks, this operation is done in the Fokker machine directly from the ground supply by means of a pump which is a fixture on the machine. Gasoline level in the tank is shown by a special type of gauge in the pilot's cockpit.



SIDE VIEW OF THE FOKKER P.II

might be obtained with the very reliable 250 hp. R. M. W. engine, using only 10 to 15 gallons of gasoline per hour.

The maximum seating capacity, including pilot, was fixed at 12, in order to obtain that speed which is the greatest asset of unserviced air transport. A load consisting of five passengers, baggage, pilot and fuel for a five hour flight is very suitable for commercial purposes at the present time, with this the machine has a radius of action of 450 to 500 miles, at a speed of 90 to 114 miles per hour, according to engine used. Long distance services are usually split up into stages falling within this radius, but the machine easily carries more fuel, for ten hours if required.

The engine is at the front end of the fuselage and drives a propeller of 6 ft. 10 in. diameter. Cooling is obtained through a new radiator of pleasing shape which fits in with the stream-line outline. Radiating surface can be regulated to suit various conditions by the pilot by means of slats.

The greatest possible safety against fire is provided by enclosing the engine compartment completely with aluminum sheet and by the steel tube construction of the engine mounting and housing. The latter was so designed that the engine is completely and easily accessible on all sides on swinging as much as the detachable wing is in necessary. By this means the labor cost and loss of time in making small adjustments or removing and replacing the whole engine are reduced to a minimum in this machine. A further feature of the Fokker system of construction is that the engine mounting is cast in shape to suit precisely any make of vertical 8 cylinder engine, up to 300 hp.

Since the front end of the fuselage and the engine is the cockpit for the pilot, whose view, forward and down, is in this position, is the most perfect mountable and excellent for orientation and landing. Fittings and instruments are reduced to the minimum desirable and are most conveniently arranged. Elevator and rudder control are operated as usual by hand and foot levers, and the ailerons by wheel.

In about the greatest safety against fatigue, etc., pressure tank has been created. The greatest tank which does not absorb any heat is made of a heavy alloy, through can pipe, or plastic pipe, when it cannot be designed in a bad landing. It is well protected from fire and so well separated from the cabin that passengers may be allowed to smoke.

The fuselage is constructed on the usual Fokker system, of steel tube, which proved so successful in the thousands of machines used in the war. It is remarkably strong, very durable and easily repaired in case of damage. The fuselage is built right up to the bottom surface of the wing behind the pilot's cockpit and contains a cabin, which is big enough to hold six people and a compartment for baggage, directly under the wing. Exhaust is effected through a door in the left hand side of the fuselage.



Detailed Description of the Fokker P.II

The seating arrangements consist of one comfortable seat, facing forward, for two or three people and two adjustable easy chairs. In case the machine is to be used for carrying mail or freight only, all the seats can be removed, leaving a clear capacity of 250 cu. ft.

Big windows, several of which can be let down, provide light and ventilation without an opening draft. The landing gear of the cabin, which can be regulated by the passengers, makes



REAR VIEW OF THE FOKKER F-11

it totally unnecessary to wear special clothing even in the coldest weather, and the danger of some illness taking on serious form is thereby cut.

There is another exit window forward, which gives communication with the pilot, and an emergency exit through the roof of the cabin.

The absence of a rear wing and landing structure of any kind provides an entirely unobstructed field of view sideways as downward, a feature which, coupled with its aerodynamic nose glider, makes this machine of extraordinary value for purposes where accurate observation is required, such as photography, surveying, forest patrol, bee-hiving, pest-control, etc. It is interesting to note in this connection that the Dutch Government Research Department uses a Fokker F-11 machine as a flying laboratory, meteorology, models, etc., being mounted some distance under the wing and radiating tubes in contact inside the cabin.

Behind the cabin, the body tapers to a vertical keel spine, to which the balanced rudder is hinged. The tail plane is fixed to the top of the fuselage and obtains its strength from a transverse stiffening beam in front and a diagonal steel strut at the rear, which connects with the bottom of the fuselage. The elevator, divided and balanced, are hinged to the tail plane. The control surfaces can all be changed in a few minutes. The tail-shed is very strongly made, of ash and strong by steel spring supports inside the body.

The chassis is very strong and consists of eight struts, two main struts and two cross. The side is of large dimensions and rubber springs. To prevent sinking in soft ground, or a bad landing through a hard tire, the wheels are double and consist of two normal runs spiked to one hub.

The plane is, as mentioned above, entirely built on the cantilever principle, has a span of 52 ft. 5 in., a chord of 18 ft. 4 in., at the body, and 8 ft. 10 in. at the tip, and is 25 ft. 4 in. at the center. It is built with two immensely strong box spars, on which solid three-quarter ribs are fixed with corner pieces. The covering of the whole is also of three-quarter ribs, which stiffen the wing as a whole and depressure with the main fabric covering which is easily damaged and deteriorates quickly.

The calculated factor of safety on the spars is 7, the load tests have proved higher with the normal ribs, etc., the factor is raised considerably, so that there is actually a bending load factor between 8 and 10.

The great thickness of the wing, its mass capacity and the fact that the ribs divide it up into a great number of rectangular watertight compartments, enable the whole machine to float passively submersible in the event of a forced landing in the water. In this case the passengers get on top of the waterbus through the emergency exit which is provided in the roof.

The altimeter are balanced like the rubber and elevators. The control cables are carried through the wings and the pulleys are accessible through inspection doors.

The fitting of the whole plane, directly on the top of the body, is carried out by four bolts of which the documents give a safety factor of 20. This simple attachment gives the great reliability that the maintenance of wings and body is always in order and it becomes possible the quickest possible maintenance.

The number of valuable working hours required to disassemble, often cover to a trivial knowledge of one of the many small parts, re-erect and erect in only one or two hours of the only wire-knived type of airplane, and the way in which it is subject to make mechanical, but dangerous, strains in many parts of its structure, need only be borne in mind to realize that the Fokker monoplaner, cheap in first cost and in operation, and requiring a very much smaller staff to keep it in daily use, is claimed to be the first airplane to make air transport as a paying proposition possible.

Contrary to many types of airplanes adapted for nonmilitary purposes, the flying conditions of the Fokker monoplaner have in no way been sacrificed to commercial requirements. It flies as well and only as any ordinary military purpose machine. It is very steady in the air, longitudinal stability is entirely satisfactory, but it can be maneuvered in all directions without effort. The lightness of control is a special feature, which prevents tiring out the pilot on long distance trips or in bad weather. There is no tendency to spin, and the machine tends to straighten again in position, both gliding and flying. It is also easy to steer on the ground.

The results obtained with the machine of the Fokker F-11 type on the London-Amsterdam Air Mail Line have been an extraordinary fact that most recent the entire air mail service London-Bombay, Amsterdam-Hankow, Copenhagen and Berlin will be carried out with a fleet of 1921 type Fokker F-11 machines, in which a few improvements on the 1920 type have been embodied. The great success of these machines has led to a special type constructed for colonial use, with an eye to climate, landing ground, and other conditions.

The F-11V type is a larger machine on the same basis, fitted with an engine of 400 to 500 hp. and carrying twelve persons plus baggage at a speed of 125 miles per hour.

Dimensions, Weights and Performance of the Fokker Commercial Monoplanes

Wingspan	52 ft. 5 in.
Wings area	1,200 sq. ft.
Wings chord	18 ft. 4 in.
Wings tip	8 ft. 10 in.
Wings weight	1,200 lb.
Wings fully loaded	1,200 lb.
Wings empty	1,200 lb.
Wings fully loaded	1,200 lb.
Wings empty	1,200 lb.
Wings fully loaded	1,200 lb.
Wings empty	1,200 lb.
Wings fully loaded	1,200 lb.
Wings empty	1,200 lb.
Wings fully loaded	1,200 lb.

When fitted with Daimler, Fiat, Liberty (500 H. P. Ford) or other engines, a corresponding increase in performance, without reduction of safety, is obtained.

Aeromarine U-8 180 Hp. Engine

Reliability must be the principal requirement of an aero engine, but it is not the sole criterion of aeronautical success. The question of fuel and oil consumption, which possesses comparatively little importance in the case of war as long as reliability and light weight were obtained, has a very important bearing on the economy of aerial transportation. Fuel and

The Aeromarine U-8 engine is tested, which was undoubtedly the first engine of this type to be built, was through the entire fifty-hour test without grinding of valves, staining of carburetor, or any other down beyond the removal of the valve raising for inspection, and no spark plug was changed during the entire test. That the engine should have taken no advantage of the carburetor passages provided by the Navy is a remarkable testimony as to its excellence, which is still further emphasized by the fact that the last twenty hours of the test were conducted in a half run, a most unusual performance.



ANTI-PROVULSION END OF U-8 ENGINE

oil economy are very pressing problems today, when the price of gasoline and of lubricants continues to soar skyward and the engine wars are so that the world's oil supply is likely to be exhausted within two or three generations.

There is still another aspect to this question. Every pound of weight deducted from the fuel supply adds as much to the useful load carrying capacity of an aircraft. Since power and weight constitute a paying load, whereas fuel is "expense load," the all important question of fuel economy will be fully settled.

It was with these requirements in view that the Aeromarine and Motor Co. of Kroyer, N. Y., started last year the design of a new engine which would embody economy, reliability, low fuel and oil consumption, great simplicity and general accessibility. This engine is of the water-cooled eight cylinder Vee type, with the cylinders arranged in two rows of four around a common crankshaft at an included angle of 60 deg. The cylinders are 5 1/2 in. by 5 1/2 in. The engine is rated 180 hp. at 1750 rpm. and the normal brake horsepower is 135 hp. The total weight complete with the propeller shaft is 515 lb., with the electric self-starter and generator it is 550 lb.

The first trials of this engine, called the Aeromarine Type U-8, recently took place at Kroyer, N. Y., and extended all yearlong. After having successfully passed a privately conducted forty-hour test the new engine was taken down and found in such a satisfactory condition that it was at once reassembled and put on the block to pass the very stringent fifty-hour test prescribed by the United States Navy. This test consists of eight five-hour runs and, at the conclusion of these, of a ten-hour run. Between the runs the Navy allows one and a quarter hour stops, respectively, the one-hour stop being for minor adjustments, while the two-hour stop allows for general overhaul, during which carbon may be removed, valves ground and inspection and spark plug cleaned and adjusted.



SIDE VIEW OF AEROMARINE U-8

At the conclusion of the fifty-hour test a suspension of the engine showed that its general condition was excellent, the carburetor, spark tracers of carbon on the valves and the absence of oil in the spark plugs being particularly noteworthy. The whole performance greatly impressed the Navy representatives who witnessed the test in official capacity.

Although the Aeromarine U-8 engine is rated 180 hp. during the Navy test it developed for half an hour 200 hp. at the rated speed of 1750 rpm., an excess of eleven percent over its rated horsepower. The unusually low gasoline consumption of this motor appears from the average of over 200 readings taken during the test, which works out at 8.472 lb. per horsepower-hour for all readings of the test, while the oil consumption was 0.611 lb. per horsepower-hour. This is a performance which merits a most desirable advance in the field of new engine construction.

What these wonderful test figures mean in everyday language will be clear from the following example. If the



VALVE GEAR OF NEW AEROMARINE ENGINE

Aeromarine 180 hp. engine were to increase the same amount of gasoline and oil as the famous German Mercedes of the same horsepower, the amount by which it was filled would have to carry approximately 170 lb. more fuel for its horsepower. And since the Mercedes weighs 180 lb. more than the Aeromarine U.S. the increased weight and load weight and expense load afforded by the new Aeromarine engine would enable a small airplane to increase the carrying capacity by some 14,000 letters.

"Municipal Landing Fields and Airports"

"Municipal Landing Fields and Airports" is edited and compiled by George George Whist with chapters by the Chief of the Army Air Service, the Director of Naval Aviation, and other well known men. It is a very comprehensive statement of the present need of airport landing fields in this country, and contains most useful information on their selection, construction and upkeep.

The fact that the development of aerial transportation depends almost entirely upon the existence of proper landing fields with equipment is well brought out. It presents numerous evils now in, the construction of airports and airports with roads, streets, shops and hotels, if proper terminals are built. The railroad men have rails, ties and equipment, the automobile a good road, while the only thing the airplane

needs is a good landing field. There will be no depression of road beds as in the case of the railroads and the truck. The speed of travel, however, is greatly increased. It sets forth the fact that the Government and airplane manufacturers are willing to help in establishing fields, but the cities and towns must first start the ball rolling. This is undoubtedly true, and the more the cities become aware to the importance of airports, the more will air transportation be securely practicable.

Chapter three outlines the construction of a field and how it should be located. It takes up the size of fields, their shape, and so forth. Landing places such as specified in the book should not be hard to find, and they would take care of any possible difficulties into which a place might run. Not only does the book deal with landing fields, but it takes up the necessity of airports, their proper location, equipment and personnel.

This book is a wonderful argument for commercial and pleasure flying. It refutes arguments that can be brought up against the future of the airplane. The men who have contributed to the book understood thoroughly the present needs of the airplane, and have done a good piece of elementary work for the country.

The volume which is illustrated and contains 66 pages, is published by G. F. Peckham's Sons, New York.

Airplanes and Forest Fires

By Paul G. Hedington

The season of forestry in the Western United States begins early upon the production of the first forest fire. The National Forests in California alone have over one thousand fires per year during the months of July, August and September. The success of effective fire protection is prompt detection and quickness of dispatch and attack in equipment. The present lookout system, developed during the past twelve years within the National Forests, differs in scope with the situation and the conditions of the airplane does not in any way supersede this principal method of detection, but merely supplements and augments its operation.

Locations of Lookout Stations

The primary lookout, located as they are on commanding heights within the forest, has a command of the forest fire for twenty-four hours of the day. They have, however, inherent difficulties in some, such as the topographic nature of the area, particularly if a limited forest visibility, and occasional adverse atmospheric conditions, which sometimes prevent the possibility of any detection. Under such conditions it is that the airplane can serve its function as a valuable supplementary agency.

The usual primary lookout, serving an area of 200,000 acres, may have a direct view of only a small portion of this total, in which he can detect a fire from its very inception. He may be able to detect, on other portions of the area, fires when they have reached a certain size and are out in the open. However, still other portions, particularly in the deep canyons, from which smoke can only be seen when it comes from a fire having reached a point of extreme potential danger.

Special Advantages of Airplane

The airplane flying at an average altitude of ten to fifteen thousand feet through the California Sierras has before it like an open book, canyon and hillside, dense timber and open meadow. No matter how small the fire, a keen observer can note its full character without difficulty. Two years a fire was reported by the airplane which proved to be nothing but extra large camp fire. It was essential to leave the observer to measure the character of the smoke, but a constant watch is kept.

It frequently happens that a keen observer posted at the National Forests during certain hours of the day when lookouts at the elevations can give an effective detection. The airplane observer, on the other hand, flying at higher altitudes, can see through the moderate haze with sufficient accuracy to serve the purpose of quick detection. This is due entirely to the difference of the angles of incidence of the sunlight from the point of view of the airplane observer and the lookout. The same holds true for certain conditions produced by the smoke coming from a large fire.

In times when the airplane serves as a supplemental detection agency, where primary lookouts are unable to detect fully. It would be pointed out, however, that during the poor fire season when only a limited number of lookouts have been placed the supplementary agencies largely depend upon the airplane for detection.

As pointed out, stationary lookouts give twenty-four hours' service. At best airplanes serve for a short specific period during the season of their flight. It is for this reason that with the high altitude station in California, where the airplane planes no detection, merely based on two daily trips over any given point, would not meet the needs.

Remembering by Airplane

The airplane this year had some opportunity to demonstrate its use in connection with definite missions in the detection and fire suppression. After a severe lightning storm which covered parts of Northern California, special large were made over the following morning. Lightning fires have proved to be a most good trouble. A time having been struck may succeed for a good many before a definite column of smoke of sufficient size to be picked up by our lookouts will be created. The airplane can see definite smoke from a distance of many miles.

The chief special use of the airplane this year was made in smoke fire, with some large fire burning in the northern part of California.

The Mill Creek fire in the Lassen National Forest was located by the airplane, immediately following the fire, and covered a very large area, besides numerous other fires which were located by the same means. The airplane was able to locate the fire, and the nature of the damage to keep in touch with all sides of the fire. It would have been out of the question for the plane to have covered the perimeter of the fire, as it was a three day's hard journey. A radio station was set up at the main fire camp and an air base accompanied the entire fire-line twice daily and kept the ranger fully informed by radio of the situation. The patrol work of the airplane after the fire was under control, saved a good many hundred dollars daily in the wages of fire-fighters and paymaster, for when supervision would have been difficult at best. The airplane was used extensively in transporting fire-fighters and supplies to various points of fire, saving days of otherwise lost travel time.

Transportation of Trained Forces

The Lassen National Forest had 32 lightning fires start in one rainy period within a twenty-day period, which taxed the capacity of the regular force. Large crews of fire-fighters were on hand but there was a decided lack in trained forest officers to handle the crews. Immediate action was required. Fortunately, the Stanislaus National Forest was then enjoying a period of quiet, when the "SNOT" from the Lassen; a plane was dispatched to the Stanislaus and brought back two forest officers who reached one of the large fires that were burning. This saved three or four valuable days. Two days later a break occurred in the Stanislaus, and a radio call to the return of the men. Again the plane was put into action, and the men were back on their forest the same night. The airplane is an extremely important, making it possible to mobilize and transport our trained forces all over the state. This mobility of trained men was most extremely lacking heretofore.

Detection Work

In detection the airplane has proved its place. With good trained observers a large percentage of fires can be located within a quarter of a mile. An observer flying over the Stanislaus National Forest reported to the forest supervisor, "A small fire, just started in heavy timber, burning slowly, wind N. W." When the plane lands at the base the observer reads up the forest supervisor to confirm the message.

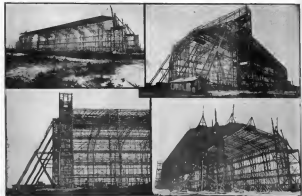
The last forest fire observed by the airplane was a forest fire, as it had been noticed by a lookout that he would have known in that general vicinity the same day. The observer is fire in his statement. When the ranger reaches the forest, the observer is enough to find it, and it is slowly taking its way northwesterly up a gentle mountain slope.

Reliability of the Service

By use of radio immediate communication is available to the observer from the forest. For the forest, the airplane is an other machine can take its place both as an accuracy and diagnosis. Taken in conjunction with the training of an adequate Army air personnel, airplane air patrol has a very definite, distinct and valuable service. Twenty million of acres of National Forest land are viewed practically every day by the observer, besides an enormous area outside, along the patrol routes. Brush and grass fire outside the forests was reported by the state forest employees in this connection. The state writer desires to record the appreciation of the Forest Service of the splendid spirit of cooperation which was at all times evident with the Air Service. The Signal Service, the state forests and many public bodies who through their contribution of money made possible the preparation of adequate landing fields and the construction thereof of improvements for the bettering of the service.

It is the writer's hope that the forest fire, better than 3,000, and for the safety, which was up into the thousands of thousands, the number of casualties was remarkably low. But the air patrol is a safe undertaking, and great good care the great attention the state forest employees, who through their cooperation, hoped that we shall have a continuation of the Airplane Forest Patrol next year and the larger use of wireless telegraphy and telephony, Forest of Airplane.

The Lakehurst Hangar Is Nearing Completion



Two building work was showing 25 inches arrived at the 193 and 1930.

View showing steel structure under construction in hangar. This view shows height of tower compared with structure in foreground.

Two building work was showing 25 inches arrived at the 193 and 1930.

View showing steel structure under construction in hangar. This view shows height of tower compared with structure in foreground.

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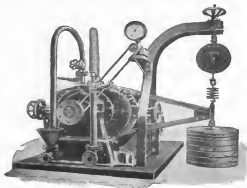
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WRITE FOR BULLETIN F-70

C. H. WHEELER MFG. CO.
PHILADELPHIA, PA.

Aeronautical Engineering and Airplane Design

By **LIEUTENANT ALEXANDER KLEMIN**

Air Service, Aircraft Production, U. S. A., in Charge Aeronautical Research Department, Airplane Engineering Department. Until entering military service in the Department of Aeronautics, Massachusetts Institute of Technology, and Technical Editor of AVIATION and AERONAUTICAL ENGINEERING. In two parts.

PART 1. AERODYNAMICAL THEORY AND DATA

Modern Aerodynamical Laboratories
Elements of Aerodynamical Theory
Sustention and Resistance of Wing Surfaces
Comparison of Standard Wing Sections
Variations in Profile and Plan Form of Wing Sections
Study of Pressure Distribution
Biplane Combinations
Triplane Combinations—Uses of Negative Tail Surfaces
Resistance of Various Airplane Parts
Resistance and Comparative Merits of Airplane Structures
Resistance and Performance
Resistance Computations—Preliminary Wing Selections

PART 2. AIRPLANE DESIGN

Classification of Main Data for Modern Airplanes; Unarmed Land
Reconnaissance Machines; Land Training Machines
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